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AVIATION

The Oldest American Aeronautical Magazine



THE *N. A. C. A.* MEETS THE *Industry*

PROBLEMS OF *Alaskan Photography*

Airplane AND *Engine* SPECIFICATIONS





Unobstructed photograph showing no damage from fire after this history of fire.

On the other side of the Steel Wall the machine shop blazed but the planes were PROTECTED in this ALL STEEL HANGAR

A BIT of hot solder, accidentally dropped into an open receptacle containing gasoline, caused an instantaneous and terrifically hot fire in the machine shop located in a lean-to of the Lorainville Field hangar at Youngstown, Ohio.

Credited in the machine shop served to increase the fierceness of the flame—greasy work benches—all soaked shavings, disk, chain and miscellaneous furniture—all of combustible material. And the fire department had a four side run, partly through the congested shop, to get there.

The steel wall separating the lean-to from the hangar prevented the passage of flames to the adjoining area with its thousands of dollars worth of flying equipment. No damage other than slightly blistered paint. Truly, striking evidence of the fire resistance of the all steel hangar!

In addition, steel hangars are fire, rugged buildings of fixed structural properties, yet portable and easily movable if moving becomes necessary. They are always weather-proof and with even casual care will last a life-time. Buildings whose characteristics have successfully weathered and withstood many fires, that have been known to withstand wind attacks up to velocities of 100 m. p. h. Other steel products are of particular interest to the aviation industry where safety and economy are paramount. Let us send you information on steel sheding, filing and office equipment, partitions, etc.—products that are as modern as aviation itself!

Trade Research Division
National Association of Flat Rolled Steel Manufacturers
211 Terminal Tower, Cleveland, Ohio

Save    with Steel Hangars

Steel sheding even under up to 100 m. p. h. winds. It was used and can be adapted to many uses around the hangar.



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This cooler, faster, more economical means of making sheet metal assemblies is very positively used by Mono Aircraft, Inc. Thousands of slow and costly tapping operations have been avoided since Hardened Self-Tapping Sheet Metal Screws were adopted. And without the least sacrifice to the security of the fastening!

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AVIATION

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Show Value, How Much, To Whom, and When?

THAT AIRCRAFT SHOWS are of value is a foregone conclusion. The extent of that value and to whom is a horse of a different hue. On one side we hear that shows are of great sales value, but there are too many of them. On the other side we hear that they are of great educational value, and that there should be more shows and better located. Regarding the question of too many shows or too few we take no sides at this time. We shall endeavor to concentrate on the value, how much, to whom, and when?

For the last three years, aircraft shows have attracted hundreds of thousands of members of the general public. Attendance records have been broken and rebroken. And in most of the principal cities of this country, Mr. and Mrs. John Citizen have become fairly well acquainted with the airplane as it is on the show floor. Therefore we may take it that aircraft shows have been of considerable educational value to the public. The public, however, did not make these aircraft shows of financial value to the exhibitors. The public paid its fifty or seventy-five cents, took a good look at things, gained some first-hand knowledge, and then went its way with purse-strings tied securely.

We of the industry used to think that the public was air-minded, and that a great profitable market was there for the taking. In time we woke up to find that we were selling our neighbor, and that the public market was but a fanciful dream yet to be realized. We now admit that we were wrong. And, as the saying goes: to admit that we were wrong is no crime, it is but saying what today we are wiser than we were yesterday. Therefore, insofar as the public is concerned, aircraft shows are of great educational value, and of practically no financial value to the industry which is exhibiting its products.

In due time, and may that time be not far distant, the public will welcome its purse-strings at aircraft shows as

it does at automobile shows now. However, at present we are confronted with the fact that the best that we can hope for from the public at aircraft shows is a few leads that may or may not result in cash sales at a later date. Yet, on the other hand, even though we can only expect a few leads, that does not mean that aircraft shows are not worth the industry's time and money. Even if shows were of no immediate financial value to the exhibitors, the advertising and promotional opportunities offered are a consideration in themselves.

However, aircraft shows are of financial value to the exhibiting members of the industry. As history shows us, millions of dollars worth of aeronautical equipment has been sold during show time. But the idea is that twenty-one and one-half per cent of these sales were made to persons and organizations actively engaged in aeronautics, and not to the public at large. Therefore, it would seem that the average exhibitor at an aircraft show has been able to cash-in in two ways. First, the opportunity to exhibit his products to the public, which will constitute tomorrow's market. And second, to sell for cash to his industry brethren which constitute today's market.

To sum it all up, the exhibitor in today's aircraft show demonstrates to the public with one hand and buys from and sells to his neighbor with the other. All of which brings up the question: when is the most profitable time for the exhibitor to buy and sell with his neighbor? Attendance figures indicate to us that he can demonstrate to the public during any of the twelve months of the year. Generally speaking, the season seems to play a minor part in getting Mr. and Mrs. John Citizen by the terrible. Aircraft show sales figures tell a different story.

For the sake of illustration, let us take the figures for sales reported at the St. Louis, Detroit and New York

shows. "We use the word 'reported' for in the aeronautics industry the word 'reported' covers, among other things, a multitude of uses!"

At the St. Louis show which held February 15th to 22nd, it was reported that sales totalled in the amount of \$2,500,000. Sales at the Detroit show held April 5th to 13th were reported to total \$1,500,000. And for the New York show held May 3rd to 11th, total sales amounting to \$750,000 were reported.

Thus it will be noted that as the year progressed sales at the shows decreased. To say that manufacturers had anything to do with it is immediately proved wrong by the fact that attendance at Detroit was more than double that at St. Louis. Whereas, sales at St. Louis were over one third more than sales at Detroit. As a matter of fact, the difference in attendance at Detroit and St. Louis is due largely to the former being a mechanical city and the latter a textile city. As far as New York is concerned, no such comparison can be made. It is a city of a little bit of everything, with perhaps a leaning toward to finance.

Therefore, placing attendance aside and regarding only the sales aspect of shows, it would seem, as exemplified by reports for St. Louis, Detroit and New York that the industry does its purchasing at the beginning of the year.

In other words, the active member already the first half show of the year, takes a good look at what competitors have to offer and then puts out his checkbook and stocks up on equipment and material for the year.

That may or may not be the case. Sales figures indicate it, however. Whether a show near the end of the year would result in an even greater sales figure is something that the future will prove or disprove.

At present an aircraft show is of educational value to the public, and of advertising and promotional value to the exhibitors. What financial value it has for the exhibitors is gauged by the purchases of the brother members of the industry. And figures indicate that the purchasing power is greatest at the beginning of the year.

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Runoff

COMPETITIVE air touring as an annually recurrent feature of the aeronautical calendar had its beginning in the United States, but for the energetic and descriptive term that heads this paragraph, the "airmail-flight" we are indebted to Germany. It carries an aroma of casual and companionable wandering, quite distinct from the businesslike atmosphere of the Ford Tour. The casualness is more in the name than in the thing itself, for competition is quite as keen on the other side of the Atlantic than here, but if consideration

of the same requires us to consider the two types of tour side by side we get some suggestions for a contest of a new type.

The Ford Tour is primarily a trial of transport planes, notwithstanding its having been run for three out of the last four years by open cockpit machines. Even this, in fact, compete primarily as light transports or mail-carriers, with no definite stipulation for passenger accommodations beyond that for the pilot and with a score proportioned to the weight that they can carry through the tour. There is no separate place and no special provision for the private-owner plane or the aerial touring machine as such, and planes designed primarily for commercial service in that class have, when they have entered at all, been on the whole strikingly unsuccessful.

The International Challenge for Light Planes goes near to the other extreme. Though entries will come with the backing of manufacturers, they are aside in the nature of an aero club instead of a factory. No plane weighing over a thousand lb. empty is allowed to compete. The rules are designed to insure a test of the qualities of touring craft as seen by the private tourist, and the race is not to the swift, for first and second places last year went to German monoplanes of very light wing loading, high aspect ratio, and well under a hundred horsepower.

The two things cannot well be combined. They call for different costs, different schedules, and different formulas of award. If the Ford Tour is to continue in its present guise there should be no expectation of securing the participation of the light and comparatively slow sport planes. They stand equally in a need of a competitive tour, and they should be given one of their own—with a special class for genuine private owners, not connected with the aircraft industry and flying their own machines.

It may take time to assimilate the idea of such an event and so get it started, although it need not and should not take long with both National Air Tour and European experience of several years past to draw upon. In the meantime it occasions no great regret to see the entries for the International Challenge again skimming America unrepresented.

There are a number of original American types which are legally qualified to enter and which, we believe, make a good showing. Laying the purely sporting element and the inherent interest of an international contest aside, there are concrete and practical reasons for having at least one American machine on the scene in future, even though it is now too late to do anything about 1935.

The slowing down of sales at home has forced our attention on the export market. The sales of American airplanes abroad and not be limited to the western hemisphere. Admitting that for the present it is out of the question to invade the territories of the great aircraft-

manufacturing countries even with a purely commercial product, there remains the Scandinavian and Baltic countries. These remain also the possibility of finding licensees to build from American designs. International licensing need not be a unilateral process. The De Havilland Company has found responsible firms to take license under its designs in at least two leading industrial countries, and our manufacturers might do quite as well. Furthermore, in course of time the private purchaser in France and Great Britain as well as in Chile and Norway is going to insist on getting the best available value for his money. It proves transactions, it is only very rarely that patriotism takes leading preference over self-interest. If the American aeronautical industry runs its affairs with as much intelligence as has gone into the design, production, and selling of American automobiles, the way to parallel in some degree the triumphs of Ford and General Motors has open before it.

All these prospective purchasers have their eyes fixed upon the Light Plane Tour, and a success there would be of immense value not only to the individual factory concerned but to the whole American industry. If an American machine could be selected, either by a committee or through a preliminary trial on this side of the water, which would have a good chance of victory it would be worth the while of the whole industry to contribute to a fund to send that one over as a representative of what American design and workmanship have to offer.

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Fireman Save My Plane

THOSE WHO think sprinkler systems are the most efficacious enemies of hangar fires, and those who hold an opposite point of view, got together in Washington recently and deliberately set fire to the corners of a wooden hangar to see for themselves what the sprinklers would do in the way of controlling the situation. It may be summed up as a contest between the power of fire, when fed by high test gasoline and doped nitrate, and the power of water as delivered via various types of sprinklers.

Experts of the Bureau of Standards and members of the Fact-Finding Committee in charge of the tests studied the aspects of every blaze with the aid of instruments—such as those indicating temperature at the roof of the hangar, rate of flow of water, and pressure in the pipes, etc. It is too early yet to hear their decisions on the value of sprinklers in hangar fires but without having to wait for their findings, we can say that, as a result of our observations, we would prefer to park our plane in a hangar equipped with sprinklers, and if we owned a hangar we would prefer having it safeguarded by the same method.

In other words, it is as it appears that the long controversy between the aviation industry and the fire underwriters as to the relative value of sprinklers in hangars has been won by the latter. To the industry's contention that water will not extinguish burning gasoline, we must agree. To the underwriters' contention that while the gasoline does continue to burn it is restricted materially from spreading to cause certain greater damage were the sprinklers missing, we cannot also and assert that this is well worth knowing about and applying.

This does not by any means rule out other means of fighting hangar fires. For instance, there are many hangars so situated as relation to the available water supply that it would be impossible to install a sprinkler system. Again, though sprinklers are installed there may well be used in every fire for these other types of equipment to be brought into play, to supplement the water flow. A hangar full of planes presents an extremely potent hazard and one would be entirely justified in applying every means, automatic and others, against the outbreak of such a disaster.

Leading up to this "trial by fire," the underwriters long ago presented to the industry through the latter's representative organizations, a code of hangar construction—of high order and including sprinklers—on which the insurance rates would be based. The industry was anxious for lower rates and wished to build hangars as inexpensively as it considered reasonable. The underwriters declared a maximum position on the construction methods, insisted on sprinklers and declared that rates could not drop below a certain point until its objectives had been complied with. The industry declared sprinklers were unnecessary, cost too much anyway, and in general it rebelled.

The upshot of several attempts to adjust differences was that the recent act of simple tests of hangar fires in which sprinklers—the chief bone of contention—would be used. The matter was placed in the hands of a fact-finding committee composed of representatives of all sides of the question and headed by Col. Harry H. Blier of the Aeronautics Branch. The Bureau of Standards co-operated, the tests were held on the Bureau's grounds and its experts assisted in the tabulation of data. Altogether, the final reports should be as authoritative and fair as any investigation into both sides of dispute could well be.

The question was an extremely important and delicate one to decide. Its settlement will influence construction of airport buildings to a marked degree. These buildings are increasing in size and cover the aspects of buildings used in other industries so, after all, who shouldn't they be subject to the same safeguards generally practiced in them? While the presence of quantities of high test gasoline complicates the situation, it has been quite strikingly demonstrated that sprinklers may be counted on for much protection.

PROBLEMS OF *Aerial Photography* IN ALASKA

By
LIEUT. R. F. WHITEHEAD, U. S. N.
Photographer Officer, Alaska Mapping Expedition



The first view of the Alaskan Aerial Survey Expedition. The Mt. McKinley, Alaska.

THE ALASKA Mapping Expedition which during the summer of 1929 mapped 13,000 sq miles of southeast Alaska employed four OL-8A airplanes equipped and suitable for mapping flights. Two planes were used for mapping, one for oblique photographs and the fourth as a standby. While these planes have not the visibility desired for mapping, otherwise they performed functions satisfactorily for this type of work.

Each mapping plane had a crew of three, pilot, navigator and photographer.

It was the pilot's duty to keep the plane at the mapping altitude of 10,000 ft., to keep it level, both laterally and longitudinally and also to keep a constant air speed.

The navigator directed the pilot along the proper flight line, signalled the photographer when to start and stop along photographs and kept a record of the time and place of starting and stopping of the runs. The navigator drove the pilot much as an individual would drive a horse. The navigator secured lines from his cockpit to the pilot's arms and by pulling on them directed the pilot on the proper course. Small lights were rigged in both the pilot's and navigator's cockpits in order to indicate when an exposure was being made. These lights were flashed on by the photographer five seconds before the picture was being taken, and off after exposures had been completed. The average interval between exposures was 15 sec., thus giving about 10 sec. in which to direct the pilot on a new course in case the plane was off the flight line.

The photographer had at his disposal the making of pictures when directed to do so by the navigator, keeping a record of exposures made during the run, and making exposures often enough to allow for a 70 per cent

In the December, 25, 1929 issue of AVIATION there appeared an exceptionally interesting article dealing with the work accomplished, and the personnel and equipment of the 1929 Alaska Mapping Expedition. We take pleasure in publishing a second article on that most important and highly successful aerial venture. Lieutenant Whitehead relates the problems encountered in the work, and places particular emphasis upon the maintenance difficulties on cameras operated in cold temperatures.

directional overlap. This overlap, or rather, the interval for the overlap, was obtained by noting the area covered in the view finder, the lines in the view finder being spaced to indicate the overlap between successive exposures. This interval, of course, varied according to the ground speed and the altitude of the plane above the territory being photographed. In Alaska where the altitude varies from sea level on the shore line to 8,000 ft.,

the exposure must be carefully watched. The shortest interval used was 7 sec., and the longest 32 sec.

The camera, of course, must be kept level and crabbled to allow for the angle between the course and the track of the plane.

Prior to each mapping flight, the mapping team would look over the charted area to be mapped. The best available charts or maps were used for flight charts—either the Coast and Geodetic Survey to the scale of 1:300,000, or the International Boundary Commission to the scale of 1:250,000. Flight lines were drawn 1/4 miles apart and were parallel at right angles to the shore line. Spacing of 1/4 miles was to allow for lateral overlap of 30 per cent.

The purpose of placing flight lines at right angles to the shore line was to take advantage of the Coast and Geodetic Survey coasted points. As some of the flight lines extended over 30 miles with practically no land marks for reference, it required careful plotting and expert navigation to keep the plane on the proper flight line, especially so when strong cross winds were encountered.

SUCCESSFUL MAPPING can be accomplished only with the greatest co-operation between pilot, navigator and photographer. Had one or the other failed to carry out his specific duties the mapping photographs would have been practically worthless.

In order to take advantage of any area that was clear of clouds, each plane was given a flight schedule with at least three choices in different localities. These choices were listed in order of preference and the navigator of the mapping plane could shift from one choice to another

to take advantage of the most favorable weather conditions. The flight schedule also directed how each plane should proceed from and return to its base. This was necessary in order to have accurate knowledge where to search in case of a forced landing of any mapping plane.

Two 4-lens cameras (T-2), two Fairchild K-3A cameras, four Stanssens developing reels and other general photographic equipment comprised the material for the Alaska expedition. The T-2 4-lens camera were used for all mapping photographs.

Following the arrival of the Gannet and the Navy Lungs at Ketchikan on May 24 everything was in readiness for mapping. On May 24 the weather was not clear enough for mapping flights, but was clear enough to dash to the mapping altitude of 10,000 ft. On that occasion the opportunity was used to test the cameras, check wiring and familiarize pilots, navigators and photographers with their duties.

On the first air tests, one of the T-2 cameras proved to be unsatisfactory, clouds being made by 30 exposures taken during a two hour flight. In the other two cameras the film tore after a few exposures. Examination revealed slight holes in the rollers, which together with the film being brittle due to low temperature, probably accounted for the casualty, as they later proved to be satisfactory after rollers had been smoothed down with crocus cloth.

On May 30 the first mapping flight was made over Kruz Island. The film torn in one of the T-2 cameras after 34 exposures. No reason could be found for this other than that the film was very brittle on account of the extreme low temperature at the mapping altitude, the temperature being minus 6 deg. F. In the other mapping

compress the shutters of B and C lenses aimed at intervals. The plungers of all shutters were examined and then the plungers and linkages were oiled with grease cloth and thoroughly cleaned. After this they functioned satisfactorily.

AIRBORNE FLIGHTS in early June indicated that the shutters were not working as satisfactorily as they should, both cameras were taken to the Peconic Long Neck pier, where a test was made to determine the action of low temperature on the shutters and plungers. The cameras were mounted and operated as in aerial mapping operations, exposures being made at regular intervals and the temperature lowered to 10 deg. F. After 40 min. at this temperature the plungers became sluggish and after an hour the shutters failed to function, or functioned only intermittently. The cameras were then taken to a warmer room and after 30 min. in a temperature of 50 deg. F. the plungers again worked satisfactorily. Then all plungers were removed and carefully oiled with grease cloth to give them more clearance in the lens bearings. Following this the cameras were tested again with the increased clearance and the shutters of both cameras performed perfectly for over an hour at 30 deg.

Before each mapping flight all plungers were removed from the shutters and both the plungers and linkages were thoroughly cleaned. In addition to this a heavy plate was mounted to fit over the base of the camera to keep dirt or any small particles from getting near the plungers or linkages. This plate was removed after the plate had left the water and the camera latch opened. It was this our practice to operate the shutters about every 30 sec. from the time of take off until ready for mapping in order to prevent possible freezing of shutters while climbing to the mapping altitude. Even after all these precautions were taken, occasionally one or more of the shutters of both cameras failed to function.

Before undertaking such mapping operations, camera shutters should be tested for at least an hour at a temperature to which they will be subjected during operations. It is difficult, except between June to note if the shutters are functioning properly. At the present time the first knowledge of shutter failure is obtained usually only after the film has been developed, fixed, washed and placed in the drying rack where the drying rack where the film can be examined for stains. It would be a distinct advantage in saving film and time and expediting mapping operations if the camera had an attachment which would release to the photographer at once when any shutter ceased to function.

The rolls upon which for the T-2 mapping cameras were one round film bracket one screw driver, an open end wrench, a string switch, 2 spot levels, and one roll of adhesive tape.

In order to determine exactly when the exposed sections of the film had been wound off, a small mirror

was manufactured and installed at an angle to the dial. This permitted the photographer to see from his seat the white line indicated when the exposed film was wound off. An instrument was usually made at rather short intervals and the photographer was very busy keep the camera level and watching overlap, this dial should be located so that the photographer can note when the film is re-wound without changing his position.

In order to compensate for the crash of the plane and thus make the photographs with respect to the flight course or to avoid crash into the photograph, the camera was rotated about its vertical axis by sliding the trimmer on the circular metal bar, but the camera latch permitted no more than a 5 deg. crash thus no account of the camera being striking the sides of the camera latch.

On first reconnaissance the expedition carried 40 rolls of hyper-sensitive panchromatic film, 380 ft. long and 6 in. wide and 30 rolls of hyper-sensitive panchromatic film for the fourth chamber of the camera. This film was 105 ft. long and 6 in. wide. Later 33 large and three small rolls were supplied of which a total of 11 rolls remained at the end of the survey. Fifteen rolls of hyper-sensitive panchromatic film 75 ft. by 9 in. were used to make oblique photographs. In addition to this 6,000 ft. of negative defense panchromatic film in rolls, 250 ft. long were re-exposed and 3,000 ft. in 100 ft. rolls. Two thousand feet of positive stock was also used. This film proved satisfactory for motion pictures.

Since the T-2 mapping film was delivered in rolls 380 ft. long, special developing reels were requisitioned. Each reel holds approximately one-third of a roll or 120 ft., and the complete roll of mapping film was developed in three cuttings. This size reel was convenient to use and proved very satisfactory.

AIRBORNE TESTS by 32 ft. holding 380 ft. of mapping film or 1,000 ft. of motion picture film resolved by a small electric motor was installed on the large and proved satisfactory. An Eversharp professional printer was used, using both AC or DC electric current. Due to the fluctuation of electric current supplied to the large, a gaudy reflector was substituted for a Cooper-Hewitt and the proved satisfactory under the conditions.

Considerable difficulty was experienced with static on the film. A careful record of temperatures was kept and



The Base House where Army Engineers at San Diego Naval Air Station were the victims of the expedition.



Seeking the temporary house at Ketchikan for a survey flight.

a careful effort made to determine the cause but with very little success.

As far as temperatures were concerned mapping operations were carried on at various degrees the lowest recorded was on May 30 when the temperature at 10,900 ft. altitude was minus 6 deg. F., and the highest on August 5, with a temperature of 40 deg.

From May 30 to July 11 satisfactory light for mapping was available between 8 a.m. and 4 p.m. At the latter 1 p.m. this being stopped on August 5 to the period between 10 a.m. and 2 p.m. After August 11 until September 8 lightning shadows reduced the time limit for mapping to the hours of 11 a.m. to 3 p.m.

The average duration of mapping flights was from three to four hours, this depending on the distance of the territory to be mapped from the base. On clear days between May 25 and August 5 each mapping plane made two flights, but after that time only one mapping flight per plane could be made. The average time to unload, clean the plungers, reload and install cameras was about one hour. Had additional cameras been available, time could have been saved by having the spare camera loaded and ready for installation as soon as a plane returned from its first flight. This would be a valuable saving for future expeditions in case of camera country, as the ships arrived for mapping are few and far between.

One cannot estimate in terms of dollars the value of the results of this expedition. It will be years before the total worth of the results can be determined. Only one experienced in conducting topographic mapping by ground methods in such a country as western Alaska can appreciate the valuable aid of the mapping parties. Aerial topographic mapping by ground method is well nigh impossible, at least within the limits of commercial cost.

Since the method of taking the pictures has been discussed, possibly some explanation of the manner of using them will be of interest. Customary to the general belief,

the finished prints are not joined together to make "mosaics," but are used in a modified plane-table method, known as the radial-line method, to determine the positions of the chosen points. Lines are drawn from the station points of two or more pictures to objects common to both, and the intersections of these lines determine the positions of the objects just as on the plane table. The station points of the pictures are the images of points on the ground above which the airplane was flying at the instant of exposing the film. They are the control points of the 11 pictures from airplanes that were in a horizontal position when exposed but are points displaced from the centers of pictures that were inclined to the horizon plane and when so displaced are in error by as much depending on the amount and direction of the tilt of the plane.

The importance of maintaining a horizontal position for the B negative when the exposure is made therefore becomes apparent. It is necessary to establish the positions of the station points on the compilation sheet in order to use them for locating additional points, and this is done substantially as follows: Three or more points whose positions on the ground have been determined by one of the usual surveying methods must be identified on the first or each series of photographs. These known points serve to determine the scale of the map, the amount of the bias of flight, and the position of the station points of the pictures. This last procedure is accomplished, as in the three-point method of plane-table surveying, by drawing radial lines from the station point to the images of the known points on the pictures and shifting these lines over the plotted positions of the known points on the compilation sheet till the lines pass through the plotted points.

Furthermore, the known points are common to the first two sets of photographs they serve also to orient the second set with reference to the first by a similar application of the three-point method. Combining this process the difficulties is able by choosing suitable points common to the overlapping series of pictures to plot the positions of a flight or more accurately, to plot from the successive sets of overlapping pictures the positions of as many common points as he may need or be able to identify.

AS THIS WORK is done on a sheet of tracing paper, not on a picture mounting and yields a series of strips representing individual flights which in turn must be assembled into the final sheet. The separate sheets are not necessarily on the same scale and when of different scales, are adjusted to one another by the known control points. Then control for each series of photographs is obtained by the known points of the United States Coast and Geodetic Survey. This method of tying the pictures gives the positions of stream courses, roads and similar features, but does not make it possible to draw contour lines and construct a topographic map. Other methods must be used if topography is to be represented. Such maps are the basic ones which the information collected by geologists of the Geological Survey is correlated, and they are absolutely essential to a complete investigation of the geology and mineral resources of a region. It is as yet only estimated as to how valuable these photographs will be to the geologist in connection with his field work and other work as correlation material previously obtained, but it is believed by those who have given it consideration that their value will be great.



Fig. 1—Aluminum gas tank showing how corrugated tops and vertical and horizontal corrugations are used to take care of vertical expansion and contraction.

By W. M. DUNLAP

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THE METHODS of joining metals are of fundamental importance, as a structure is dependent upon the strength of the joints. Aircraft welding is the application of the general principles of welding to a particular field. A knowledge of the methods of joining metals is essential for every mechanic, draughtsman and engineer engaged in the construction and repair of aircraft.

In the early days of the industry aeronautical engineers, for the most part, were opposed to welding in the structural parts of aircraft. However, the aircraft industry is expanding with great rapidity, and old methods are being discarded and new ones adopted. The higher load factors required by the more powerful modern engines, as well as other considerations, have resulted in a trend toward the all-metal plane.

At the present time the welding of aluminum in aircraft construction ranks second in importance to that of steel. The principal application at present is the welding of gasoline and oil tanks fabricated from sheet aluminum. This is the preferred method of construction in the case of military aircraft, as well as in commercial airplanes, where the maximum pay load is the goal. Welding aluminum tanks solves the difficulty of producing a gasoline-tight joint without the use of cement and fillets which are always of questionable durability. The welding of aluminum is not difficult and can be easily mastered by the aircraft welder. However, in the welding of aluminum, as well as steel, every precaution should be taken to insure a sound weld.

The fuel tanks used ten years ago were generally made of iron plate, although in a few cases brass and copper have been tried. These tanks were satisfactory, in that they were durable and easy to repair. The weight, however, of a fuel tank made of these metals was a disadvantage. The logical method of reducing the weight was to change to a material with a lower specific gravity. Aluminum sheet comes in this class. A comparison of

the weight per gallon of gasoline capacity indicates that that of the welded aluminum tank is only a little more than 50 per cent of that of similar tanks made from the other commercial sheet metals. It is possible to carry between 2½ and 3 gal. of gasoline per pound of aluminum tank. Gasoline, oil and water tanks for the aeronautical power plant, due to their involved shape, must be strong and light, and contain baffle plates to prevent slopping of the liquid.

A fuel tank should be designed to withstand an internal air pressure, without buckling or injury, of not less than 3.5 lb. per sq. in. Large flat surfaces in the shell of the tank should be avoided unless they are well stayed by bulkheads and stiffened by corrugations. The stiffness of aluminum is less than that of steel or brass; therefore, the thickness of the aluminum sheet should be greater. In the average type of tank, a thickness of 0.051 in. is usually satisfactory for tanks up to about 300-pal. capacity. Cylindrical tanks may be constructed with few stiffeners except in the heads, which should be dished, covered or corrugated. A cylindrical shape, however, is generally not economical of space in the airplane, hence, tanks with a square or rectangular cross-section are more common. The flat sides of the tank must be reinforced by stiffening with corrugations and by the use of bulkheads. Fig. 1 illustrates the use of concentric rings and vertical and horizontal corrugations. These corrugations also assist in allowing for the thermal expansion and contraction resulting from the heat of welding and thus prevent distortion and warping of the shell. The welding line is applied only to the area at the seam. The result is that the skin expands and, unless this movement has been allowed for in the design, a vehicle will form. When cooling, the reverse action takes place, but the wrinkles will not pull out. The result is that a heavy strain is placed on the welded seam. The welder cannot prevent the expansion and contraction of the metal. The designer must take care of this.

Bulkheads which are used not only to reinforce the tank, but also to prevent slopping or surging of the liquid, contain a large number of lightening holes, and openings at the bottom to permit complete drainage of the tank. Bulkheads are riveted to the shell of the tank and the tops

Welding IN DESIGN

of the rivet heads are welded on to make these gas-tight. Fig. 2 shows the welded rivet heads. The rivets should not be too large as that would cause difficulty in welding the heads. A ¼-in. rivet is large enough for sheet up to 0.051 in. in thickness. The welding of the rivet heads should be done with care, as the building up of too much material on top of the rivet head will make a lump which causes stresses in the sheet at the edge of the weld and is apt to start a crack. The corners of the tank should be rounded so much as practicable.

THE EXISTENCE of fittings such as filler necks, gasoline gauge fittings, and pump fittings introduce complications in the fabrication of tanks. These fittings should be manufactured of aluminum tube or aluminum alloy castings in order to reduce weight. Aluminum fittings cast and should be welded to the shell of the tank, but, to prevent warping, a bead should be placed concentrically

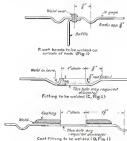


Fig. 3—Welded rivet heads and fittings

With the tendency toward the use of metal in aircraft construction the problem of welding aluminum and its alloys becomes one of prime importance. Considerable research has been conducted to determine the possibilities of this form of fabrication in the laboratories of the Aluminum Company of America. In the accompanying article Mr. W. M. Dunlap of the metallurgical division has treated in a general way the possibilities of welding aluminum sheet and has also touched briefly on the welding characteristics of the strong alloys. While he does not advocate welding of the alloys for highly stressed structural members, he suggests a number of possibilities in this field which have not yet been utilized.

with the fittings or the fitting should be attached to an offset or reinforcement in the tank. Fig. 2. The fitting used for the fitting should be of the same chemical composition as the sheet from which the tank is formed. Such fitting may be welded to the sheet very easily. The castings contain, principally copper and silicon as the alloying constituents and may vary widely in composition. The aluminum-silicon alloy castings are preferable for welding, even though the aluminum-copper alloy castings may have somewhat better mechanical qualities. Care must be exercised when welding heavy castings to the shell of a tank. The wall of the tank must be unobstructed or corrugated to take care of the expansion of the sheet. If this is not done, the welding heat will buckle the sheet upward at the edge of the casting. When the metal cools this buckle will not flatten out. Therefore, rivets are set up in the weld, which, added to the strain of vibration, may cause cracks to develop in service.

Another feature to be avoided in the construction of tanks is the attachment of two castings so close together that the welded seams intersect. The fault indicated there can be corrected by separating the castings a sufficient distance so that each one can be executed with a depression in the sheet, or can be rounded out on a separate



Fig. 4—Welded joints in aluminum sheet

piece of sheet aluminum, and this welded to an end-member on the sheet.

Several types of joints in welded sheet are shown in Fig. 3. These joints, if properly welded, will give strengths equivalent to the strength of the aluminum in the unwelded condition. This holds true no matter what temper of aluminum sheet is welded or how much excess metal is deposited in the bead, as the area immediately adjacent to the welded seam will be annealed or at least partially annealed by the heat of the torch during the process of welding.

Tanks must be thoroughly cleaned after welding to remove all traces of the welding flux. Generally speaking, the oxide coating on aluminum is beneficial in that this very tenacious, highly refractory and chemically resistant film tends to protect the metal beneath from the action of the atmospheric elements. However, this oxide must be removed from the welding zone if a satisfactory joint is to be obtained. A flux that will adequately remove the oxide film, will attack the metal beneath. Aluminum differs from most other metals in this respect. Most metallic oxides adhere rather loosely to the base metal, leaving a tendency to scale off. Furthermore, the oxides are attacked more readily by certain chemical reagents than the metal itself. When iron or steel is pickled in an acid bath to remove the rust, the oxide coating is completely removed without the metal being attacked to any appreciable extent.

If any trace of the flux is permitted to remain on the welded article for any appreciable length of time, the aluminum will be corroded. A very efficient means of cleaning the flux from welded aluminum tanks is to wash in hot water to remove the bulk of the flux, then immerse in a hot 10 per cent sulfuric acid solution for a short period of time and rinse thoroughly in hot water.

After washing, the welded tanks should be tested for leaks. This may be done by connecting an air hose and completely submerging the tank in water, forcing in air

and looking for leaks, which would be indicated by bubbles of air rising to the surface of the water. In case the tanks are so large and awkwardly as to make submergence impracticable the joints may be painted with soap suds and examined carefully for bubbles when the internal pressure is sufficient. The tanks should be tested at an internal air pressure of from 35 to 60 lb per sq in., or more, depending upon the size, shape and design of the tank. A recovery column is usually used for indicating the pressure as mechanical pressure gauges are not sufficiently accurate at these low pressures.

Welded tanks should preferably be supported in cradles or slings rather than rigidly attached to the aircraft structure. Metal strips with a protective lining between the strip and the tank make good supporting members. The lining may consist of a strip of annealed pure aluminum, or a strip of felt, impregnated with paraffine oil, or asphaltic base paint. Sawdust or composition material containing free alkalies or chlorides should never be used as there is danger of the tank being corroded by these materials. Welded aluminum tanks for use in airplanes and other aircraft operating near salt water should be well protected on the outside by painting with aluminum paint. A finish coat of enamel may be applied over the aluminum paint if desired.

There are many places after this point where it might be of advantage to substitute welding for riveting. For example, fanning, wing position stiffeners, cowling, pilot seats, fuselage and other parts that do not go to form highly stressed structural members, and where a reduction of strength in the vicinity of the seam is not objectionable, may be easily welded. There are also places where electric resistance spot welding could be used to advantage. Ammunition carriers on parent or fighting planes could be spot welded. It is probable that in the near future wing parts such as aluminum, batten and struts will be so designed as to permit their being assembled by spot welding. The baffle plates for gasoline

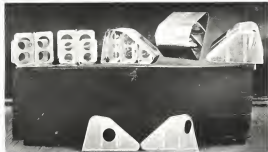


Fig. 3: (above) Butt joints fabricated from aluminum. Fig. 3: (below) Wing tanks fabricated from aluminum.



Fig. 4: Welded aluminum tanks for C-47 aircraft.

has received an air quick. The area adjacent to this has been raised to the annealing temperature and is intended to move outward. The area which has received an artificial aging by the heat of the torch. A portion of this area is, in all probability, over aged. Beyond this area, the sheet has not been affected by the heat of welding. The total area of the sheet affected by the heat of welding will probably extend from one to two inches beyond the welded seam on either side. If the structure is heat treated subsequent to welding, the affected areas in the wrought metal will disappear. However, detrimental in the cast condition, even when heat treated, does not develop the high mechanical properties of the wrought material, hence the strength and ductility of the cast metal in the seam will remain low. Ductiveness as welded has a tensile strength of from 38,000 to 45,000 lb per sq in., and an elongation of about 4-6 per cent in two inches. By hammering after welding, the strength may be raised to 46,000 to 47,000 lb per sq in. and the elongation to 6-10 or 7-10 per cent. By hammering and heat treating after welding, the strength may be raised to 50,000 lb per sq in. or more, and the elongation raised to 8-10 or 9-10 per cent. However, hammering and heat treating a welded structure is not always practical. In order to obtain good corrosion resistance, the material should be water quenched as it comes from the furnace. This would induce warping and buckling in the welded structure.

and all tanks could be assembled by spot welding instead of riveting.

The welding of strong aluminum alloys has not reached a state where it can be recommended for highly stressed structural members in aircraft. The strong aluminum alloys can be welded very readily. However, it must be borne in mind that the metal in the seam is in the cast condition and that the material in the vicinity of the seam is reduced in tensile strength. If we stop to analyze the conditions prevailing during the welding process, we will find that the following is probably true: The metal in the seam proper has melted and reoxidized, hence is in the cast condition. The metal immediately adjacent to the cast area has been raised to the heat treating temperature by the heat of the torch flame, and

Automatically controlled electric heat treating furnaces are being built at the present time. It is not beyond the realm of practical possibility that, in the not distant future, light aluminum aircraft structures will be assembled by welding, slung in rapid runs, heat treated and water quenched. The writer can visualize methods that the engineer of yesterday would have considered extremely radical, but which the engineer of tomorrow will accept as standard practice for assembling light aluminum aircraft structures.

THE N.A.C.A. MEETS THE Industry

*Technical Progress Reviewed and Problems Presented
at Fifth Langley Field Conference*

By LESLIE E. NEVILLE
Technical Editor of Aviation

ANOTHER milestone in the progress of the National Advisory Committee for Aeronautics has been passed in the form of the Fifth Annual Aircraft Engineering Research Conference held May 23 at Langley Field, Va., and once more the viewpoints of the commercial and research engineers have been compared.

From the standpoint of the commercial engineer it would be a mistake to say that there was any one outstanding feature of the conference. It is probable that each visitor had a different opinion as regards the most important work of the N.A.C.A. However, many of the pertinent problems of the aeronautical engineer are recurring rather than new.

Probably the underlying theme of the research work of the N.A.C.A. is a tendency toward full scale and full flight measurements. The laboratories are constructing larger wind tunnels, the variable density tunnel is beginning to show its value and the flight test section is develop-

ing new and improved instruments to measure the characteristics of an airplane subjected to various maneuvers under free flight conditions.

As usual the arrangements of the N.A.C.A. to make the trip comfortable and instructive were carefully planned and the entire schedule was followed strictly without deviation. The more spectacular part of the program was repeated the following day for the benefit of the service men stationed at Langley Field. This was made necessary because of the somewhat restricted capacity of the laboratories and the officers' club where the speaking sessions were held. As the service men were more interested in the inspection of the laboratories it was thought desirable to provide this feature for them exclusively on a separate day. The industry was well represented as a result of the care exercised by the committee in extending invitations to prospective visitors.

As in the case of former conferences, there were three

general divisions in the session. The morning period at which Dr. Joseph S. Ames presided as his usual incompressible manner, was devoted to a brief review of the year's work by the heads of the subcommittees engaged in various research programs. Following this the visitors were divided into four groups for inspection of the various laboratories where spectacular experiments were set up to illustrate the work.

The third period, in the afternoon, was provided to permit visiting engineers to present their general problems and offer suggestions as to the future work of the laboratories.

Space limitation prevents a verbatim report of the

conference at this time, but the complete minutes will be distributed by the N.A.C.A. in the near future. In the interim we will attempt to outline briefly the general content of the three sessions. All of the information discussed at the morning session, as well as the facts gathered by experiments shown during the inspection tour, have been published recently or is scheduled for early publication in the form of the *Summary Reports, Memoranda and Notes of the Committee*.

The session on aircraft was opened by Dr. Ames and conducted by Henry Ford, engineer in charge of the Langley Memorial Aeronautical Laboratories.

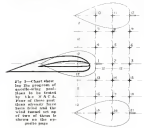
Dr. Ames spoke briefly of the Committee's work and purpose, and enlarged the late Capt. L. M. Woodham who had presented the Packard Diesel engine as its first cross-country test flight at the previous conference. Dr. Ames also told the assembled visitors of the new wind tunnel and water channel, both of which are in the course of construction on the laboratory property. The new wind tunnel will be rectangular in shape, having a 30-ft by 60-ft throat, and will have two propeller fans driven by a 6,000-hp. engine which is expected to provide a stream velocity of 100 to 120 m.p.h. The water channel is to be more than 2,000 ft long, 25 ft wide, and 12 ft deep. It is expected that models will be propelled through it at a maximum speed of 60 m.p.h.

After an address of welcome by Commandant West of Langley Field, the first speaker was Mr. Elton W. Miller, chief of the aerodynamic branch, which is one of the two subdivisions of the laboratory.

Mr. Miller outlined concisely several of the more important problems of the aerodynamic section. One of these is the program devised to investigate the effect of engine nacelle position on the lift and drag of an



Visitors and guests at the Fifth Annual Aircraft Engineering Research Conference



ordinary services such as the Clark Y and R-4F-4, but off in efficiency beyond one-half of the speed of some of the three-seater fighters of the late and the other services earlier than that point. A species of building phenomenon has been observed on small aerol services at high speed. He further stated that the maximum ordinate had been moved to the center in tests and it was discovered that those 95 per cent of the speed at which the section then produced proved more efficient than either the Clark Y or R-4F, and at 18 times the speed of speed it was found to be 15 per cent more efficient.

Mr. B. Chown suggested that since the heating system has been shown to give satisfaction, a careful study at this time of the heating system of the airplane in the attitude should be made in order to obtain the full advantage of the wall. He further stated that the Wright company had been working on superheating and learned that a large temperature rise limited their work on the ground. He requested a curve plotting highest possible temperature ratio against water temperature for a standard fuel. Mr. Henry M. Cruise of the Bureau of Standards replied that this would be a difficult problem and that, while the Bureau of Standards could predict the general form of such a curve as absolute figures were in possession at the moment. He suggested a collaboration of the Bureau of Standards and N.A.C.A. to obtain this information. Mr. Chown received this information regarding the inlet temperature and the location and intensity of hot spots would probably be as helpful in his work. Dr. H. C. Debus of the Bureau of Standards told of an work on case of flame propagation and mentioned that twelve factors have been found to enter into the problem of detonation, the most recent of these being luminosity of charge.

MAJOR R. H. PLANT next spoke, congratulating the N.A.C.A. on its work and stated that he had two important problems, the nature of which he would not dredge.

Charles Ward Hall stated that the progress in covering experiments had by far exceeded his expectations and that he had two additional problems: the problem of new studies of cabin leakage shapes, and the other a method of eliminating burning over the back of the fuselage and enflaming over the pilot's cockpit in case of open flames. He also asked further investigation of the effects of plating in engine in the process it normally occupies in a flight test.

Dr. Charles G. Abbot of the Smithsonian Institute suggested study of the problem of jet propulsion and Mr. Eustace N. Jacobs, director of variable density wind tunnel, stated that such propulsion had possibilities for very high speed craft. He commented, saying that a large number of the aircraft were carried off by the jet and that it would be necessary to increase the momentum of the jet, which had been accomplished to a slight extent. Dr. H. L. Dryden suggested that the Bureau of Standards co-operate with the N.A.C.A. in studying methods of measuring effective thrust of jets.

A desire to obtain additional information on the effect of aerodynamic deceleration rate, including those of the recently completed foreign ship, was expressed by Dr. K. Amundt and Mr. C. B. Fritzsche. Dr. Wolfgang Klumpner requested that no interest be developed to lead the torque of an engine at a point on the shaft between engine and propeller to obtain the difference between the two. He suggested that work was in progress on a link dynamometer, but that no work had been planned for

long shafts. Mr. A. E. Lawson told of the progress in the development of the engine and stated that as problems were of such an individual nature that he was not yet in a position to inform direct to the N.A.C.A.

Mr. C. N. Menzies reviewed his losses that the designer controls but 35 per cent of the weight of an airplane and said that he would recommend the elimination of some of the component parts. He requested specific information on the relationship of rubber and its use to dihedral angle, and Mr. Crowley stated that in the light test work of the N.A.C.A. he also had been referred to a relatively low point, but that the results had not yet been observed. Professor Alexander Klumpp suggested a resort to the chosen theory to solve this problem, and Mr. B. V. Kervin-Kowalewsky stated that his suggested method had been based on constant in a number of problems and that the calculations involved required only about one hour. [The reader is referred to an article on "Proportioning the Airplane for Lateral Stability," by Mr. Kervin-Kowalewsky in the *Aeronautical Engineering Section of AVIATION*, January 19, 1929.—ED.]

Mr. L. C. Williams commended the Committee upon its construction of the new model house, and Mr. W. H. Miller suggested that the flight test work be supplemented by a theoretical study of pressure distribution.

ON the outstanding talks of the afternoon was that by William B. Stout in which he commended the Committee for its frankness in presentation of facts without bias and pointed out the advantage of obtaining negative, as well as positive results in research. Mr. Stout agreed with Mr. Menzies that some of the parts of the present-day airplane would have to be discarded in the future and suggested the possibility of reducing drag by elimination of leading edges.

Mr. George Loring, charged with Messrs. Minnigh and Stout, saying that airplanes could not be simplified and made attractive to the individual citizen as an example the addition of brakes and starters. He suggested research on starters and variable pitch propellers, as well as the shape and disposition of front windows of cabin planes. He suggested low static wall, and some slight center and commended the N.A.C.A. for its use of Langklopp and known to determine the air flow over a fin used in one of the cooling studies. He suggested that this method be used to determine the effect of airplanes on radio and fire.

Mr. C. T. Foster asked that the aerolic relation tests be extended to the hypersonic and that some attempt be made to determine the relation of propeller diameter to nacelle diameter. He stated that the increase in performance obtained by use of a ground engine was frequently greater than that reported, and he suggested a study to this relationship. He also suggested an investigation of the aerodynamic and cooling properties of modern engine nacelles.

The second problem was answered up by Mr. W. B. Mayo, who said "Name in the product of efficiency." He suggested study of propeller noise, possible changes in propeller systems and elimination of external noise center as possible. He also stated that he was interested in large airplanes having multiple engine nacelle layout, and the relation of nacelles to support wing to be told, he pointed out that planes were limited in use usually by engine size.

Following the closing remarks, winter visitors were free to inspect in detail any of the experiments or airplanes at which they were particularly interested.

THE SECOND NATIONAL

Airport Conference

A Review of the Committee Reports Read During Three Day Meeting of the Aeronautical Chamber of Commerce

By CHARLES H. GALE

Assistant Editor of AVIATION

DURING the twelve months elapsed since the first national airport conference at Cleveland in May, 1929, the attention of the airport interests has swung from the constructional aspect of the airport to operation and problems of refining the equipment already installed. It was on such matters that the airport and airport equipment men concentrated during the second national port conference held at the Hotel Statler, Buffalo, on May 14-16, last week, under the auspices of the Airport Section, Aeronautical Chamber of Commerce of America.

The session of this second "clearing house" for airport problems indicated that plenty of perplexities must be apparently they are chiefly those connected with growth and progress rather than with hard times. Whatever the state of the industry may be experiencing, it appears that for the most part the interests of bringing the airport facilities of the country to a high degree of usefulness is working steadily onward.

A feature that year was the provision of an association, to each of which was assigned one of the outstanding airport problems for consideration. Each member of the conference attended at least one of these concurrent meetings and theoretically participated in the preparation of the group's report to the conference as a whole. The committee meetings occupied most of the first day of the conference and the reports were of the second day. This method made possible covering many subjects quite instantly, an accomplishment otherwise impossible without several more days being added to the program.

Another feature was the display of exhibits of fourteen manufacturers and the Aeronautical Chamber, in space adjoining the conference room and the exhibition of airport equipment staged in the Statler Hotel. Most of the exhibits were on the last afternoon and evening of the program.

While the attendance was a little disappointing in the light of the promise of the conference as the most ar-

port conference of war of the year, there was nevertheless an excellent representation from all parts of the country and Canada, with 159 individuals actually registering. It is interesting to note that 39 of these represented established airports or airport committees while the balance represented manufacturers of equipment or specialists in service to the airport, publications, etc.



Victor Yarnes, vice-president of the Airport Section, Aeronautical Chamber of Commerce, who presided at Friday morning session.

As far as the issue represented in the concurrent meetings and the reports of these committees—discussed in the following—the questions which seem to be causing the present country today are the ways and means of handling aviation fuel and lubricants and the desirable policy to be adopted with respect to local handling charges and water taxes. In the same connection it is made in several instances.

displayed a wide diversity of opinion, there being strong advocates in both sides of the question whether or not there should be a state tax levied to the improvement of airports within the state, whether or not there should be a levy by the individual airport on the fuel handled by a transport company exclusively for its own planes, etc.

No action was taken by the conference in respect to these problems except to vote that it be referred to the executive committee of the Airport Section of the Chamber for further investigation. While on the subject of fuel, it should be mentioned that at the luncheon meeting Thursday, May 15, Maj. E. F. Aldrin, of the

Standard Oil Co. of N. J., read the resolution of the Advertising, Sales and Edits Committee of the Fuel and Lubricants Section of the Chamber to the effect that its members, which includes about all the principals of companies, shall not give airports free gasoline and oil, or pay money, grant special discounts or furnish free advertising for exclusive rights at an airport or make any donation of money, money or products in connection with a flying project, or carry direct accounts or display advertising for the benefit of airport operators. This group also accepts the code of marketing practices adopted by the American Petroleum Institute.

This action is extremely important since it will eliminate many undesirable competition practices and abuses of the "free" or discounted concessions often made to airports.

Among the unrecorded but very valuable by-products of the conference were the informal meetings of the airport managers and of their several sub-committees during a number of sessions. These were:

1 That the Airport Section of the Chamber be subdivided to five groups to include Managers, Public Officials, Owners and Operators, Members, and Airport Engineers.

2 That joint advertising to the Chamber be identified as such by its managers, this to be displayed on the field.

3 That two committees be appointed, one to include airport and air transport men for the discussion of port charges and discounts and the other to be a special group for the study of fire and safety measures.

THE CONFERENCE got under way officially at the luncheon at the Hotel Statler, Washington, June 14. The keynote address, prepared by John S. Elymore, vice-president of the Pennsylvania Railroad and read by his representative, explained the tremendous expense of providing adequate terminal facilities for aircraft engaged in the business of transportation. Such facilities in the national world are engaging the attention of some of its best minds and airports merit the same consideration. In launching the Transcontinental Air Transport Inc. he said, five weeks of the early went into ground facilities.

The rest of the day was given over to the holding of committee meetings in which the subject areas of airport development and operations were discussed under the leadership of a designated authority in the field. Each committee drew up certain recommendations or resolutions which were introduced in reports made to the conference immediately in the following day both morning and afternoon.

THE FIRST of the committees to make reports on their individual meetings held on the first afternoon of the conference, presented their briefs at the evening session on Thursday, May 15.

T. R. Newell, chief engineer of the Cleveland Me-

tropol Airport, led off with a report from his committee on surfacing and drainage problems, the two problems over which many an airport operator is tugging his hair. The report made the following important points:

1. There should be no such thing as seasonal availability of the surface of an airport. It should be suitable for use under all conditions and at all times of the year.

2. The basic load and smoothness requirements for landing areas are the same, no matter what the type of surface may be.

3. Subsoil condition is vital and the big point is obtaining proper subsoil or provision of adequate drainage. A thorough study of the drainage condition by an expert should be carried out before any road drainage project may be launched.

4. Graveling is another fundamental consideration, taking into consideration the maximum grade allowed by the Aeronautics Branch, and the influence of large average rainfall and the possible maintenance of hard surface.

5. Grades of 1.5 per cent are preferable. With runways or paved areas, a grade of 0.2 per cent to 0.4 per cent provides ample run-off.

6. The basic principles in drainage are: (a) Provision of sufficient outlet. (b) As few turns and sub-sinks as possible on account of expense. (c) Long laterals are usually preferable to shorter ones, where the grade permits their use, because of saving in installation of more expensive manholes. (d) Open ditches are to be avoided.

7. The herringbone system of drains with parallel laterals or the gridiron system of a series of long parallel laterals discharging into a receiving main from one side only were recommended as best for airport work.

8. Spacing of drain lines as follows was recommended:

Soil	Recommended	As a per cent of dry weight
Clay	14 in. apart	10 in. apart
Siltstone	16 in.	12 in.
Gravel	20 in.	15 in.

E. A. Johnson, president of Dayton Airport was the next on the stand. He reported the recommendations of his committee, which had taken up the matter of standard accounting systems for airports. Some time ago Frank J. Webb, assistant treasurer of the Aeronautical Chamber of Commerce and J. W. Ellingerberger, president of Universal Accounting Systems, of Washington, were asked by the Chamber to compile a chart of accounts and an explanation of them, together with a simple personnel organization chart, for the consideration of the committee and the conference.

Submission of the chart of accounts and the organization chart formed the basis of Mr. Johnson's report. Both charts are to be available to the operators through the Chamber. It is not necessary to construct on the individual basis in the chart of accounts but the point should be made that this matter of airport accounting is an extremely important one which is receiving expert and exhaustive study. In the coming months the above men will continue their services and prepare sample record forms and other accounting papers for the complete coverage of all phases of the financial transactions of an airport.

AVIATION

May 24, 1939

AVIATION

May 24, 1939

SOME INTERESTING POINTS were brought out by Maj. J. John Berry, manager of the Cleveland airport, in his report on the recommendations and comments of the lighting committee. They were as listed:

1. All airport lighting now is on an on-standby basis. It is largely accidental in nature, that is, it is based on methods that happen to work but are not necessarily the most desirable.

2. The public must have confidence in port night lighting as it does for highways, industrial operations, etc.

3. Football fields usually are lighted by systems averaging approximately 48 lux of lighting load as compared with the average of less than 15 lux for airports, which are many times larger in area. Our airports are not built enough.

4. Perimeter boundary marking rapidly is being made obsolete through general tendency in runway markings. Major Berry expressed the personal opinion that obstructions about an airport should be floodlighted and that the landing plane should depend solely upon its wing lights to pick up the ground.

5. Incandescent type of floodlight has advantages over arc lamps in ease and economy of operation but because of difficulty in controlling glare and because of lack of "parab" is the lamp, as compared with the arc, its development has lagged.

6. It was recommended that airport lighting manufacturers and representatives of the industry in committee sessions through the year study lighting problems with the view to close co-operation with the Aeronautics Branch.

In the discussion following the report it was brought out that the General Electric and Westinghouse companies are co-operating with the Aeronautics Division of the Department of Commerce in the development of a beacon which may be installed on high towers for use as a night or day aid.

The General Electric type is to be operated by induction and the Westinghouse by capacitor coupling. Experimental installations are expected about July.



Maj. John Berry, who presided at meeting, opened Wednesday, May 15.

THE REPORT of the hangar construction committee was read by Capt. R. H. Hight. This committee was headed by an engineer at National Air Transport. The main points:

1. The general trend is for larger and more permanent hangars. Average width at openings or new hangars is about 120 ft.—about 45 ft. wider than the average two years ago. It is a question if this is large enough, however. The committee recommended a minimum width of 100 ft., depth of 100 ft. and clearance of 30 ft. at first class airports.

2. A careful study should be made of the fire hazard and it was recommended that a committee be appointed to function continuously through the year for the purpose of gathering information on new fire and co-operating with the National Board of Fire Underwriters in setting up standards for airport fire protection.

3. Day lighting of hangar interiors should be studied.

4. Heating remains another big problem. A minimum temperature of 50 deg. was recommended. It was emphasized that insulation of all openings should be attended to carefully. An exhaustive study of doors was recommended.

5. The size of hangars depends considerably on the location of the building. When the doors face the landing area an economical depth is about 120 ft. When the door is at right angles to the field, the depth may be 240 ft. When planes may use both the side facing the field and the side away from the field the depth may be 120 ft. and the length as long as desired.

6. The most satisfactory floor is of concrete designed for the load to be carried, with a 1 in. steel crosshatched metal top, with proper drainage and expansion joints.

In a discussion of airplane systems following the reading of the report it was brought out that the insurance rate at Newark Metropolitan Airport had been lowered from \$4.45 per \$100 to \$0.23 per \$100 with the installation of sprinklers.

CONVENTIONAL use of the growing sources of airport income, occupied the attention of a group under the chairmanship of W. Sagar Green, manager of the Central Airport, Camden, N. J. His report stated that the group represented the following opinion:

1. That municipal airports need to be careful about maintaining a high degree of operations as far as expenditures were concerned, including nearly transportation, terminal and associated industrial facilities.

2. That non-revenue expenditures in any case should be separated from flying activities.

3. That where there was a danger of controversy running wild under outside operators at some point adjoining the airport it was better for the city to maintain control itself.

4. That hangar lessees should have the privilege of owning and operating their own gasoline and oil equipment for their own use but should not be allowed to transport.

5. That fire prices should be standardized and should not be exclusive to any one fire company unless perhaps, when handled by the airport management.

6. That the best way to handle restaurant concessions is to secure a guaranteed monthly minimum with a percentage bonus beyond that. Vending on the field should be in the hands of the restaurant concessionaire.

THE CONVENTION on small town airports, championed by Frank L. Bernierke, manager of the airport at Beaumont, Tex., was the following recommendations:

1. That the Aeronautical Chamber undertake the provision of airports in small communities through contacts with civic clubs and others, and that the small airports committee be made a permanent body in the Chamber's organization.

2. That there should be an airport in every community, no matter how small, for the following reasons: (a) It is a bad habit, correct city planning to prepare for future developments along all lines. (b) It is economically justified on opening up potential sources or avenues of revenue. (c) Because it automatically makes accessible the nearest and fastest form of transportation.

THAT TRAVEL in fact becoming a major problem at many airports was brought out in the report, and ensuing discussion, of the committee assigned to study the subject. C. W. Short, Jr., manager of the Tulsa

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Manassas Airport, read the committee's report and its recommendation that a standard code of air traffic regulations be established at all airports. As a first step in that direction, a set of rules was suggested.

After much discussion this movement was temporarily shelved, pending the outcome of the study of the same problem, now well under way, on the part of the Aeronautics Branch. The government's work in this field has been under way for many months and a report should be forthcoming before long.

Discussion developed over some of the subject points in the above-mentioned suggested rules and it was apparent that the regulation requiring a left turn after take off at all times was a particularly objectionable item. Many points it was shown, are so stated that such a left turn would create a real hazard in accident of direction. It was evident that such a regulation throughout the country would be confusing, to say the least.

SIGNAL AND ADVERTISING in the airport operation have to do with getting the airport "known" to all possible clients—from the public to manufacturers who might locate there—to the end that as income may be as large as possible. Obviously this is of more concern to the commercial airport which has been launched as an investment than it is to the municipal project, which is of a good deal of a public utility. However, income is of concern to both, and it is particularly so in these days when the ledger is more apt to show red ink than black.

The subject was taken up by a committee headed by Robert F. Craig, assistant to the president of Fairless Airport, Kansas City, Kan. The report emphasized the necessity of applying every means of sales effort, to attract public and professional patronage of the port and discussed the many methods of making these applications.

These methods are quite generally recognized. They include selling to a prospective airplane purchaser the convenience that a certain airport is equipped to give him better service than a civil field, creating the reputation of adequate air-guards for handling passengers and regularizing the crowds of spectators, selling a manufacturer or an airline company the idea that a certain airport affords facilities better than another field, etc.

In other words, the committee brought to the attention of the conference the fact that most airports or ports, having a financially hard time must resort to all the refinements of the art of salesmanship to attract the income needed for successful operation. The information starts with the airport manager and is passed on to its final possibilities, will include every man working at the field. It is an example of how the force of healthy competition may operate in our airport system to bring about highly developed plants which otherwise would not be realized, at least so soon.

H. H. Mills, manager of Brainerd Field at Hartford, Conn., reported that the committee on service and storage, of which he was chairman, recommended the same principles for determining costs should apply to commercial and municipal airports alike. That bargains should afford about a 10 per cent net profit, and that the kind of storage and service should be standard throughout the country. It was felt that charges for that service must vary with the location on account of differences in building and operating costs.

It was urged that consideration be given to the mixing

of service at airports in the same manner parts are rated for general facilities, lighting and use of landing area. A standard price of \$0.30 per gallon for gasoline and \$0.25 per quart of oil were proposed.

On the evening of the second day, the conference impact was held. The speakers included Capt. St. Hubert Wilkins, Charles S. (Casey) Jones, M. B. Rodgers, vice-president of the United States Lines and Senator J. Kenneth Webb, chairman of the New York state aviation committee.

Most city governments are floundering around in confusion over what policies should be adopted regarding many aspects of municipal airport development and operation. Col. C. O. Sherrill, former city manager of Cincinnati, declared at the morning session, Friday, May 16.



W. R. Gregg, Weather Bureau, who made an closing session, Friday, May 16.

Their uncertainties involve around such questions for instance, as whether bargains on municipal airports should be built by the city or operating company, what is the best type of management for an individual airport, whether or not flying schools not approved by the government should be permitted to operate at a field in competition with

approved schools, how far a city should go in the direct control of such activities as sale of fuel, operation of restaurants, hotels, etc.

The program for all this lies in the formation somehow of a national policy for the guidance of cities interested in airport development. Col. Sherrill declared. He urged a nationwide survey of the entire municipal airport situation, affirming his belief that there must be a better way to run the airports now perplexing the average city port. He suggested the possibility of individual title or chain ownership or operation of terminals after the fashion of many railroad and motorway terminals today.

AT THE Friday morning session W. R. Gregg, chief of the Aeronautics Division, Weather Bureau, Washington, stated freely his conviction that the responsibility of making the decision regarding carrying on a fight in doubtful weather should not be shifted from the pilot to the weather bureau representative. That question of responsibility has been rather a problem one along the airways recently and this statement from Mr. Gregg indicated that no change is contemplated in the long-standing policy of the Bureau in that regard.

Taking everything into consideration the conference was extremely north wide. With nearly no looking could have been supplied. I believe, by the introduction of a larger element of representation from the Aeronautics Branch and have a more alert local consumer. The former would have been able to contribute much toward the filling out of the complete "round table" and the latter could have told us the majority of the commercial local needs. On the whole, the Aeronautics Chamber of Commerce deserves considerable credit for the program it staged.

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MANUFACTURERS' SPECIFICATIONS ON ENGINES AVAILABLE FOR COMMERCIAL USE AS COMPILED BY AVIATION ENGINEERS' ASSOCIATION. THE TABLE GROUP IS REPRODUCED BY PERMISSION OF THE AVIATION ENGINEERS' ASSOCIATION.

Engine Model	Max. Horsepower	Max. RPM	Max. Fuel Consumption (GPH)	Max. Oil Consumption (GPH)	Max. Air Consumption (CFM)	Max. Altitude (ft)	Max. Endurance (hrs)	Max. Range (miles)	Max. Speed (mph)	Max. Climb Rate (ft/min)	Max. Takeoff Weight (lbs)	Max. Empty Weight (lbs)	Max. Fuel Capacity (Gals)	Max. Oil Capacity (Gals)	Max. Air Capacity (Gals)	Max. Water Capacity (Gals)	Max. Other Capacity (Gals)	Max. Other Capacity (Gals)	Max. Other Capacity (Gals)
Continental C-12	120	2400	12.0	1.0	120	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-14	140	2400	14.0	1.0	140	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-16	160	2400	16.0	1.0	160	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-18	180	2400	18.0	1.0	180	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-20	200	2400	20.0	1.0	200	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-22	220	2400	22.0	1.0	220	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-24	240	2400	24.0	1.0	240	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-26	260	2400	26.0	1.0	260	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-28	280	2400	28.0	1.0	280	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-30	300	2400	30.0	1.0	300	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-32	320	2400	32.0	1.0	320	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-34	340	2400	34.0	1.0	340	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-36	360	2400	36.0	1.0	360	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-38	380	2400	38.0	1.0	380	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-40	400	2400	40.0	1.0	400	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-42	420	2400	42.0	1.0	420	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-44	440	2400	44.0	1.0	440	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-46	460	2400	46.0	1.0	460	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-48	480	2400	48.0	1.0	480	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-50	500	2400	50.0	1.0	500	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-52	520	2400	52.0	1.0	520	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-54	540	2400	54.0	1.0	540	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-56	560	2400	56.0	1.0	560	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-58	580	2400	58.0	1.0	580	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-60	600	2400	60.0	1.0	600	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-62	620	2400	62.0	1.0	620	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-64	640	2400	64.0	1.0	640	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-66	660	2400	66.0	1.0	660	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-68	680	2400	68.0	1.0	680	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-70	700	2400	70.0	1.0	700	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-72	720	2400	72.0	1.0	720	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-74	740	2400	74.0	1.0	740	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-76	760	2400	76.0	1.0	760	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-78	780	2400	78.0	1.0	780	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-80	800	2400	80.0	1.0	800	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-82	820	2400	82.0	1.0	820	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-84	840	2400	84.0	1.0	840	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-86	860	2400	86.0	1.0	860	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-88	880	2400	88.0	1.0	880	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-90	900	2400	90.0	1.0	900	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-92	920	2400	92.0	1.0	920	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-94	940	2400	94.0	1.0	940	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-96	960	2400	96.0	1.0	960	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-98	980	2400	98.0	1.0	980	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10
Continental C-100	1000	2400	100.0	1.0	1000	10000	10	100	100	1000	1000	1000	10	1.0	10	10	10	10	10

May 24, 1936

Specifications of American Commercial Airplanes

Excludes sale shares with assumed Trust liabilities—ESTATE does not assume responsibility for the shares sold

[illegible]

May 24, 1950

Specifications of American Commercial Airplanes

Inclusion with plans with Approved Time Certification—WYATSON does not assume responsibility for the hearing plan.

[illegible]

Specifications of American Commercial Airplanes

NOTES AND CORRESPONDENCE—All letters and comments on articles must be sent to the Editor, c/o the American Society of Human Genetics, 11 Dupont Circle, N.W., Washington, D.C. 20036.

[illegible]

Specifications of American Commercial Airplanes

Including only plans with Approved Type Construction—BIFAPRO does not become involved in the design of the building system.

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Specifications of American Commercial Airplanes

Reseller only clients with Approved Type Certificates—ATA/AVR does not assume responsibility for the success given

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Specifications of American Commercial Airplanes

Rediffusion only deals with Approved Type Certificates-ADDITION does not assume responsibility for the license given

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PERSONNEL

The **ANENT P** team chief of the division of personnel in the Library of Congress, has been awarded the medal presented annually by Villanova College for distinguished service in the advancement of science.

HAROLD A. SEER, recently assigned manager of the manufacturing division of Nicholas-Bellamy Aircraft Co., has been named to become developer for Boeing NB-3 helicopters on the West Coast with headquarters in Los Angeles. Jack A. Seer, former general manager and general sales manager of Salsburg Aircraft & Engine Co., takes charge of sales of the Boeing NB-3 at the factory.



JOSEPH N. SCHAEFER, newly appointed general manager of Tuff-Gardner Motor Corp., has been appointed general manager of the commercial division and the Federal Supply Co. of American Airlines, Inc. The Supply unit of S. F. Corp. has recently been merged with Universal Airports, Inc. at St. Louis Municipal Airport.

HYRSH A. BARNET, recently assistant in the position of Berlin-Hoyer Aircraft Co., is now associated with Metallurgical Laboratories, Inc., Philadelphia, Pa., as quality supervisor.

J. D. GOSWAMI has been elected president of the Nebraska chapter of N.A.A. **JOHN McGUIRE** is vice-president of E. Gossens, a security contractor.

DR. DAVID VAN KAMMEN, aviation administration professor at the University of Austin, Germany, is to teach helicopter design materials at California Institute of Technology.

LEROY EDWIN M. FARRIS has resigned as chief pilot for Southern Aircraft Corp. to participate in the formation of a new airline to operate from Little Rock, Ark.

THOMAS A. BYRNE has been appointed replacement director of United Aviation School, becoming chief of the school.

JERRY KERNOWSKI is now chief instructor at Omaha Aviation School. **J. P. GOSWAMI** is sales manager.

WALTER J. SORRELL has been made assistant director of J. C. Rice Flying School, San Diego, Calif.

GARY C. BRYAN has been named assistant manager of the Twin City Airport, Duluth, Minn.

W. W. WISNER has been promoted to the position of airplane sales manager for Carlin-Wright Flying Service. G. Wisner has been placed in charge of ground and machine schools operated by the Los Angeles branch. E. H. Wisner has been assigned as the manager of the Flying Service. D. H. Wisner has been made assistant manager

at Grand Central Air Terminal, Cleveland, Ohio, and is also purchasing agent for the Flying Service there.

WILLIAM ROBINSON has been named district superintendent of the St. Paul-Dulles line of Western Air Express. **MALCOLM B. WHITE**, a director of Colonial Airways System, has been elected president and chairman of the board of American Maryland Airlines Co., New York, N. Y. **MAJ. VERNOR CHAPMAN** has been elected vice-president and other directors elected are: **MAJ. GEN. JAMES P. O'BRIEN**, **R. M. AYER**, **GEN. J. C. CAMPBELL**, **E. DELL**, **EDWARD CONNELLEY**, **WILLIAM B. MANN**.

A. BAYNE has taken charge of the C.A. office in Knoxville, Tenn. **RAY** is in charge.

HAIMER HENNING, Washington representative of The Aviation Corp., is now assistant to the president. He will remain at the Washington office.

C. T. BLACKBURN has resigned as aviation advisor of the Tampa, Fla., Chapter of Commerce to take another position. In making his resignation, he said that commercial affairs in Tampa were at a standstill and the airport situation was closed up.

S. S. C. MONROE and **W. B. HONOR** have been elected co-presidents of the Canadian Aviation Association. **MR. HONOR** has been named to be the national business and Mr. **MONROE** has been named to be the president of Bellair.

WALLACE G. GOSWAMI is now flight manager of Alaska-Northern-British Columbia Airlines with offices in Victoria, B. C.

E. A. DUNN, consulting engineer for Lewis Air Products Co., was elected president of the American Welding Society at its recent annual meeting.

JOHN E. FARRIS has been named general sales manager for Eberly Aircraft Corp., Idaho City.

DECEMBER 27 has been made field assistant at Lambert-St. Louis Airport.

C. R. BARNETT has resigned his position as non-president and works manager of Kentucky Aircraft Corp.

HAROLD P. KIRBY has been promoted to the position of plant superintendent at Canada Flight Co., Cleveland. Other manufacturers of Liau-Wild aircraft and Studebaker aircraft.

MAJ. G. E. BROWNE has been named **MAJ. RALPH B. BROWNE** in flight instructor at the Ford Personal Group, Sebring, Fla. **MAJ. BROWNE** has been assigned to duty in Washington, D. C.

PHIL GEE-BEE Canadian Branch

SPRINGFIELD (Mass.)—General Bredford Aircraft, Inc., the city's sole manufacturer in the GEE-BEE line are planning to establish a branch factory in Canada and Zealand D. General Bredford, who has been making an extensive tour of Canada since his return from the United States, is now turning out more than a week.

AERONAUTICAL CALENDAR

May 23-24 All-North Air Transport Convention, Cleveland, Ohio.

May 24-25 National Air Transport Convention, Cleveland, Ohio.

May 25-26 National Air Transport Convention, Cleveland, Ohio.

May 26-27 National Air Transport Convention, Cleveland, Ohio.

May 27-28 National Air Transport Convention, Cleveland, Ohio.

May 28-29 National Air Transport Convention, Cleveland, Ohio.

May 29-30 National Air Transport Convention, Cleveland, Ohio.

May 30-31 National Air Transport Convention, Cleveland, Ohio.

May 31-1 June 1 National Air Transport Convention, Cleveland, Ohio.

June 1-2 National Air Transport Convention, Cleveland, Ohio.

June 2-3 National Air Transport Convention, Cleveland, Ohio.

June 3-4 National Air Transport Convention, Cleveland, Ohio.

June 4-5 National Air Transport Convention, Cleveland, Ohio.

June 5-6 National Air Transport Convention, Cleveland, Ohio.

June 6-7 National Air Transport Convention, Cleveland, Ohio.

June 7-8 National Air Transport Convention, Cleveland, Ohio.

June 8-9 National Air Transport Convention, Cleveland, Ohio.

June 9-10 National Air Transport Convention, Cleveland, Ohio.

June 10-11 National Air Transport Convention, Cleveland, Ohio.

June 11-12 National Air Transport Convention, Cleveland, Ohio.

June 12-13 National Air Transport Convention, Cleveland, Ohio.

June 13-14 National Air Transport Convention, Cleveland, Ohio.

June 14-15 National Air Transport Convention, Cleveland, Ohio.

June 15-16 National Air Transport Convention, Cleveland, Ohio.

June 16-17 National Air Transport Convention, Cleveland, Ohio.

June 17-18 National Air Transport Convention, Cleveland, Ohio.

June 18-19 National Air Transport Convention, Cleveland, Ohio.

June 19-20 National Air Transport Convention, Cleveland, Ohio.

June 20-21 National Air Transport Convention, Cleveland, Ohio.

June 21-22 National Air Transport Convention, Cleveland, Ohio.

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June 28-29 National Air Transport Convention, Cleveland, Ohio.

June 29-30 National Air Transport Convention, Cleveland, Ohio.

June 30-1 July 1 National Air Transport Convention, Cleveland, Ohio.

July 1-2 National Air Transport Convention, Cleveland, Ohio.

July 2-3 National Air Transport Convention, Cleveland, Ohio.

July 3-4 National Air Transport Convention, Cleveland, Ohio.

July 4-5 National Air Transport Convention, Cleveland, Ohio.

July 5-6 National Air Transport Convention, Cleveland, Ohio.

July 6-7 National Air Transport Convention, Cleveland, Ohio.

July 7-8 National Air Transport Convention, Cleveland, Ohio.

July 8-9 National Air Transport Convention, Cleveland, Ohio.

July 9-10 National Air Transport Convention, Cleveland, Ohio.

July 10-11 National Air Transport Convention, Cleveland, Ohio.

July 11-12 National Air Transport Convention, Cleveland, Ohio.

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July 29-30 National Air Transport Convention, Cleveland, Ohio.

July 30-1 August 1 National Air Transport Convention, Cleveland, Ohio.



Operators Meeting With Post Office Dept.

WASHINGTON (AP)—A post-processed conference between representatives of the principal air passenger and mail lines of the country and the Postmaster General was held here May 19. Attention on the last day centered on the problems relating to the subsidizing of the Western Mail as to its air transport of mail by passenger planes.

Conference and passenger loads in one of the features of the Western Mail which confers the following program: "Where the air mail moving between the designated points did not exceed 2,000 cu ft., or 250 lb. per trip, the Postmaster General may award to the lowest responsible bidder, who has carried and operated an air transport service on a fixed daily schedule every 10 days or less than 250 mi. and for a period of not less than 60 days, prior to the advertisement for bid, a contract at a rate not to exceed \$4.00 per mile for a weight of 25 cu ft. or 250 lb."

Although strict secrecy has been demanded for certain of the conference, it is believed that the principal officials were permitted to discuss whether to compromise on their demands or fight for contracts on the basis of competitive bidding.

The Post Office Department has announced that the operators have agreed to prepare a map of the United States showing in detail a comprehensive plan for an air mail and passenger network. A committee of eleven headed by William P. MacCann, Jr., was appointed to work out the details.

Canadian Firm Expanding Lines

MONTREAL (AP)—A sizable expansion of an organization is being carried out by the Canadian Air Lines of Canada. In the transport field plans are making for a direct service between Ottawa and the city of Quebec, an airfield distance of about 250 mi., operated by Transcontinental Airways, a subsidiary. There is a minor side to it, a possible alliance between the parent company and Western Canada Airways, of Winnipeg. Canadian Aerial Services has been formed to take over the scheduled and of later Provincial Airways, another subsidiary.

P.A.A. System Has Increases in Traffic

NEW YORK (AP)—Starting the first 4 days of the new P.A.A. System, says Spence, earned a total of 32,499 persons, it has been announced by J. T. Spence, head of the organization. The figures for months were as follows: January, 2,576; February, 3,495; March, 3,227; April, 2,800. The passenger load, by month, were as follows: January, 29,026; February, 41,601; March, 32,632; April, 63,228.

The Mexican Division of the system, known as the Mexican Airlines Co., experienced an increase in passenger traffic. There was a total of 60,000 passenger miles flown during the first 4 days. There is an increase in mail being sent from Latin American points, it was announced, and to meet this two Lockheed aircraft have been shipped for operation along the West Coast of South America. During the first 4 days of 1959, the system carried about 330,000 lb. of mail.

Abilene (Tex.) Air Terminal From the Air



SPRINGING new look in the building is Abilene's new port. The apron, stairs and parts of runway may be seen. The port was dedicated in the April 2 issue of AVIATION.

Research on Airports To Be Jointly Conducted

WASHINGTON (AP)—The American Research Council and the American Road & Builders Builders Association will cooperate with the American Branch of the Department of Commerce in a study of the design, construction and operation of airports. These organizations have held the project under consideration for the past 6 mo. and now are ready to proceed to prepare a comprehensive report for future guidance in the development of airports. A joint committee will be organized at once and such recommendations will be organized as they are necessary to carry on the project. A study of the design and construction of airports is a lesson that is scheduled for immediate attention.

Dr. E. E. Hines, former chairman of the American Engineering Council's committee on research, presented the plan at a session of the Council's executive board on May 14, together with a report by Council's committee on airports, of which Dr. Ralph J. Fong, head of the department of civil engineering at Lehigh University, is chairman. The on operation of the Council and the Road Builders Association has been sought by the American Branch because it was felt that the diversity and complexity of the problem involved in the design, construction and operation of airports for participation by competent non-governmental agencies in promoting the improvement of a proper method about which a wide, varied and comprehensive review may be fully established.

Theoretical Model Planned

"The policy of 'make haste slowly' will create a greater real acceleration in the advancement of aviation than one of proceeding on hastily conceived and inadequately prepared plans," says the report of Dr. Fong's committee. "To make such a study, the committee will study the desirability and accuracy for careful scientific planning and execution. With the growing popularity and increasing number of airports, airports have been some 1,500 which have been built in the United States in the past 10 years. The expenditure of vast sums of money and energy. During the past 2 yr alone, the expenditure has exceeded \$100,000,000. Besides the airports already built, there are more than 1,500 in various stages of construction or under construction."

"Some of the ports appear to be well located, designed, constructed and operated, but it is just as apparent that many have been built in an uncoordinated manner."

(Continued on page 1014)

THE BUYER'S LOG BOOK



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Elgin Panel

The Elgin Instrument Division of the Elgin National Watch Company offers the industry a new panel. Included are tachometer, airspeed, compass, altimeter, pressure gauge, fuel-meter, air velocity indicator, clock, and bank indicator.



Model air instrument panel

The cluster of the airspeed and altimeter and one piece compass, ensuring rigidity and perfect alignment. Each of these instruments have four jewels showing in their complex in the front window. Panels including other components of instruments can also be supplied as well as individual instruments. —*AVIATION*, May 24, 1935

G. E. Lighting Control

BY MEANS of a new central control system, part announced by the General Electric Co., the various lighting apparatus of an airport can be operated from one point, thus eliminating the necessity of having the lighting control at a place where the various lights themselves are visible. The control panel may be placed on a wall, table or desk, or it may be mounted on a special pedestal. The rotating lights are powered from lines placed in compound cables and each light is wired to an actual crystal controlled, providing an actual indication of the condition of that circuit. —*AVIATION*, May 24, 1935

Wind-Driven Generator

ELECTRIC Specialty Co., Stamford, Conn., has recently developed a three-conductor, wind-driven generator in connection with a storage battery for use, as a current supply for airplane radio apparatus while the plane is at rest as well as in the air. Generators are being used with two transmitters one to deliver high voltage for the radio transmitter, and the other to deliver low

voltage for charging the storage battery and to furnish power to the elements of the transmitting tubes.

The three-conductor machine, according to the manufacturer, is designed so that it may be operated from the same storage battery which it charges, and deliver full high voltage output. Thus full power may be obtained from the transmitter when the plane is on the ground and the radio engine set in operation. In this machine the high voltage commutator and winding are the same as in the ordinary two-conductor, locally current generator. —*AVIATION*, May 24, 1935

Barlow Roller Fuel Pump

THE National Steel Products Co., Dayton, Ohio, Division of Aero Supply Wile, Co., Inc., is sole distributor of the new Barlow roller fuel pump to the aviation industry. The pump built by McCord Machine & Mill Co. of Detroit, has only six moving parts—four steel steel rollers, the rotor which drives the rollers and the relief valve in the pump. The mounting design is the standard S.A.E. aeromarine mounting. A single bypass valve is used in this machine, and no plunging valves. It is said to operate in either direction of rotation, but when assembled for clockwise rotation, will not function in the opposite direction. Approximately 60 gal. of gasoline per hr. at 3,600 r.p.m. are delivered. —*AVIATION*, May 24, 1935

Airport Fueling System

"THE Aroper" manufactured by O-Bell & Barber, Springfield, Mass., provides a fueling system adaptable to any size or type of airport. It is usually installed at the flying base, where it is connected to the plane by hose. Gasoline is supplied to the pit by an electric pumping unit placed on an auto or trailer and operated from a switch in the pit. The steering wheel is operated by the foot and is so arranged that during the pit covers automatically through the pit. A hand water pump is provided to raise water jets into the pit from flooded fields.

Each pit has two compartments—the control compartment containing the motor, filter, motor switch and shut-off valve, and the hose compartment with a horizontal hose and reel which is coiled 50 ft. at base. Since the hose can be taken out either side of the pit, the fueling range is approximately 100 ft. Gasoline is raised from the pit through a filter which removes dirt and sediment, and a displacement type motor increases

TRADE CATALOGS

► *Whitlock 19 Three-Engine Lesson*. The National Institute of the Whitlock 19 three-engine lesson has just been issued by the Whitlock Aircraft Works, Yonkers, N.Y. The general design is briefly presented together with special features such as wireless equipment, heating and ventilation, standard instruments provided and night flying equipment.

► *Apex Tools*. A new catalog of the Apex Machine Co. describes in detail the various types of quick change chucks, both power and hand drive, for single and multiple spindle operations, and includes tools recently brought out, full floating and semi-floating holders, bearing tap stands and universal joint nut drivers.

► *The Madras Fire Chief*. The March-April issue of this publication of American-LaFrance and Foamite Corp. contains a section relating to airplane fires, which will be of interest to the aviation industry.

the quantity disposed. The system is only 18 in. deep, in other words any installation makes it easier to keep clean and leaves the possibility of gathering forces collecting. —*AVIATION*, May 24, 1935.

Stanley-Unishear

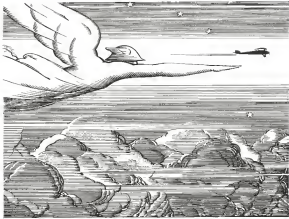
A NEW service system based upon the success of the Mighty Midget Unishers has been announced by the Stanley Electric Tool Company, New Britain, Conn. Some of the materials cut by this tool, are rubber, canvas, asbestos, celluloid, linoleum, wire mesh, sheet iron and steel. Operating from a hand socket, the shear has a capacity of 18 U.S. gauge and lighter on sheet steel, 16 U.S. gauge on aluminum and brass and at the same proportion on other materials.



A new model cutting tool

steel. The speed is up to 15 ft. per min. and depends on the loading of the operator. The constant radius of the shear is 1 in. and equipped with an 8 hp. Westinghouse Universal motor at weight 65 lb. —*AVIATION*, May 24, 1935

HARVARD WINS THE CONTEST ALOFT!



HARVARD'S flying club was the Gewinner of the \$1,000 prize for showing the most progress in student flying last year. Detroit took second, Yale, third, and New York University, fourth; with 40 colleges in the contest. The Crimson won because its 35 members all have

boats in the air, because 22 of them are licensed pilots, and because the club's sturdy "Whirlwind"-powered training plane was kept aloft for 1,155 hours as less than a year. Weight, whose engines help train most of these men, bids good luck to the 4,000 collegians now learning to fly!

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Pan American's passenger service from urban Miami to the densely settled city table



Photo by George H. Brown



SIKORSKY AMPHIBION



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Still carrying the 1000 lb. load, this mighty bomber executed manoeuvres expected only of pursuit planes. Aerobatics such as loops, Immelmans, wing-overs and inverted flying were accomplished with an ease and grace that astonished the experts assembled to witness this remarkable test. Although numerous unsuccessful attempts have been made by others, these feats have never before been performed by any airplane in any country.

Significant of the Martin Company's manufacturing dependability is the fact that it was chosen to build this great plane by the U. S. Government, which excels all other nations of the world in aircraft research. The aerodynamical design and arrangement of the Martin XT5M-1 were ably engineered by the Bureau of Aeronautics of the U. S. Navy. The structural and detail design was developed by the Martin Company.

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